Generalized Single Degree of Freedom Systems

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Generalized SDOF's

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Continuous Systems

by Rayleigh's
Method

Selection of Mode

Refinement of Rayleigh's Estimates

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Vibration Analysis by Rayleigh's Method

Selection of Mode Shapes

Refinement of Rayleigh's

Continuous Systems

Vibration Analysis by Rayleigh's Method

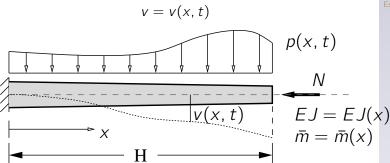
Selection of Mode Shapes

Refinement of Rayleigh's Estimates

Consider a cantilever, with varying properties \bar{m} and EJ, subjected to a load that is function of both time t and position x.

$$p = p(x, t)$$
.

The transverse displacements v will be function of time and position,



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... and an hypothesis

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To study the previous problem, we introduce an approximate model by the following hypothesis,

 $v(x, t) = \Psi(x) Z(t),$

that is, the hypothesis of separation of variables

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$$Z(t) = v(H, t)$$
 and
 $v(H, t) = \Psi(H) Z(t)$

Refinement of Rayleigh's Estimates

Timespie of Virtual Displacements

For a flexible system, the PoVD states that, at equilibrium,

$$\delta W_{\mathsf{E}} = \delta W_{\mathsf{I}}.$$

The virtual work of external forces can be easily computed, the virtual work of internal forces is usually approximated by the virtual work done by bending moments, that is

$$\delta W_{\rm I} pprox \int M \, \delta \chi$$

where χ is the curvature and $\delta\chi$ the virtual increment of curvature.

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The external forces are p(x, t), N and the forces of inertia f_1 ; we have, by separation of variables, that $\delta v = \Psi(x)\delta Z$ and we can write

$$\delta W_{\rm p} = \int_0^H p(x,t) \delta v \, \mathrm{d}x = \left[\int_0^H p(x,t) \Psi(x) \, \mathrm{d}x \right] \, \delta Z = p^\star(t) \, \delta Z_{\rm Refinement of Rayleigh's Estimates}$$

$$\delta W_{\text{Inertia}} = \int_0^H -\bar{m}(x)\ddot{v}\delta v \,dx = \int_0^H -\bar{m}(x)\Psi(x)\ddot{Z}\Psi(x) \,dx \,\delta Z$$
$$= \left[\int_0^H -\bar{m}(x)\Psi^2(x) \,dx\right] \,\ddot{Z}(t)\,\delta Z = m^*\ddot{Z}\,\delta Z.$$

The virtual work done by the axial force deserves a separate treatment...

The virtual work of N is $\delta W_N = N\delta u$ where δu is the variation of the vertical displacement of the top of the chimney.

We start computing the vertical displacement of the top of the chimney in terms of the rotation of the axis line, $\phi \approx \Psi'(x)Z(t)$,

$$u(t) = H - \int_0^H \cos \phi \, dx = \int_0^H (1 - \cos \phi) \, dx,$$

substituting the well known approximation $cos\phi \approx 1 - \frac{\phi^2}{2}$ in the above equation we have

$$u(t) = \int_{0}^{H} \frac{\phi^{2}}{2} dx = \int_{0}^{H} \frac{\Psi'^{2}(x)Z^{2}(t)}{2} dx$$

hence

$$\delta u = \int_0^H \Psi'^2(x) Z(t) \delta Z \, dx = \int_0^H \Psi'^2(x) \, dx \, Z \delta Z$$

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Approximating the internal work with the work done by bending moments, for an infinitesimal slice of beam we

$$dW_{\text{int}} = \frac{1}{2}Mv''(x, t) dx = \frac{1}{2}M\Psi''(x)Z(t) dx$$

with M = EJ(x)v''(x)

$$\delta(dW_{\rm Int}) = EJ(x)\Psi^{"2}(x)Z(t)\delta Z\,dx$$

integrating

$$\delta W_{\text{Int}} = \left[\int_0^H E J(x) \Psi^{"2}(x) \, \mathrm{d}x \right] Z \delta Z = k^* Z \, \delta Z$$

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► the shape function *must* respect the geometrical boundary conditions of the problem, i.e., both

$$\Psi_1 = x^2$$
 and $\Psi_2 = 1 - \cos \frac{\pi x}{2H}$

are accettable shape functions for our example, as $\Psi_1(0) = \Psi_2(0) = 0$ and $\Psi_1'(0) = \Psi_2'(0) = 0$

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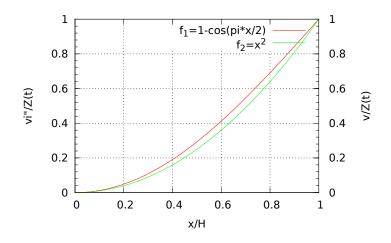
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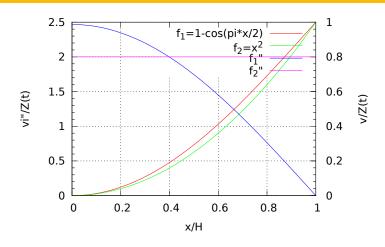




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better results are obtained when the second derivative of the shape function at least resembles the typical distribution of bending moments in our problem, so that between

$$\Psi_1'' = \text{constant}$$
 and $\Psi_2'' = \frac{\pi^2}{4H^2} \cos \frac{\pi x}{2H}$

the second choice is preferable.

Using $\Psi(x) = 1 - \cos \frac{\pi x}{2H}$, with $\bar{m} = \text{constant}$ and EJ = constant, with a load characteristic of seismic excitation, $p(t) = -\bar{m}\ddot{v}_{q}(t)$,

$$\begin{split} m^{\star} &= \bar{m} \int_{0}^{H} (1 - \cos \frac{\pi x}{2H})^{2} \, \mathrm{d}x = \bar{m} (\frac{3}{2} - \frac{4}{\pi}) H \\ k^{\star} &= E J \frac{\pi^{4}}{16H^{4}} \int_{0}^{H} \cos^{2} \frac{\pi x}{2H} \, \mathrm{d}x = \frac{\pi^{4}}{32} \frac{EJ}{H^{3}} \\ k^{\star}_{G} &= N \frac{\pi^{2}}{4H^{2}} \int_{0}^{H} \sin^{2} \frac{\pi x}{2H} \, \mathrm{d}x = \frac{\pi^{2}}{8H} N \\ p^{\star}_{g} &= -\bar{m} \ddot{v}_{g}(t) \int_{0}^{H} 1 - \cos \frac{\pi x}{2H} \, \mathrm{d}x = -\left(1 - \frac{2}{\pi}\right) \bar{m} H \, \ddot{v}_{g}(t) \end{split}$$

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Vibration Analysis

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► The process of estimating the vibration characteristics of a complex system is known as *vibration analysis*.

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- ▶ We can use our previous results for flexible systems, based on the *SDOF* model, to give an estimate of the natural frequency $\omega^2 = k^*/m^*$

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Vibration Analysis by Rayleigh's Method

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► The process of estimating the vibration characteristics of a complex system is known as *vibration analysis*.

- ▶ We can use our previous results for flexible systems, based on the *SDOF* model, to give an estimate of the natural frequency $\omega^2 = k^*/m^*$
- ▶ A different approach, proposed by Lord Rayleigh, starts from different premises to give the same results but the *Rayleigh's Quotient* method is important because it offers a better understanding of the vibrational behaviour, eventually leading to successive refinements of the first estimate of ω^2 .

Rayleigh's Quotient Method

Our focus will be on the *free vibration* of a flexible, undamped system.

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Rayleigh's Quotient Method

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▶ inspired by the free vibrations of a proper SDOF we write

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Refinement of Rayleigh's Estimates

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Refinement of Rayleigh's Estimates

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▶ the displacement and the velocity are in quadrature: when v is at its maximum $\dot{v}=0$ (hence $V=V_{\rm max}$, T=0) and when v=0 \dot{v} is at its maximum (hence V=0, $T=T_{\rm max}$,

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- disregarding damping, the energy of the system is constant during free vibrations.

$$V_{\text{max}} + 0 = 0 + T_{\text{max}}$$

Selection of Mode Shapes

Refinement of Rayleigh's

Now we write the expressions for $V_{\rm max}$ and $T_{\rm max}$,

$$V_{\text{max}} = \frac{1}{2} Z_0^2 \int_S E J(x) \Psi''^2(x) \, dx,$$

$$T_{\text{max}} = \frac{1}{2} \omega^2 Z_0^2 \int_S \bar{m}(x) \Psi^2(x) \, dx,$$

equating the two expressions and solving for ω^2 we have

$$\omega^2 = \frac{\int_S EJ(x)\Psi''^2(x) dx}{\int_S \bar{m}(x)\Psi^2(x) dx}.$$

Recognizing the expressions we found for k^* and m^* we could question the utility of Rayleigh's Quotient...

Refinement of Rayleigh's

▶ in Rayleigh's method we know the specific time dependency of the inertial forces

$$f_1 = -\bar{m}(x)\ddot{v} = \bar{m}(x)\omega^2 Z_0 \Psi(x) \sin \omega t$$

 f_{\parallel} has the same *shape* we use for displacements.

• if Ψ were the real shape assumed by the structure in free vibrations, the displacements v due to a loading $f_1 = \omega^2 \bar{m}(x) \Psi(x) Z_0$ should be proportional to $\Psi(x)$ through a constant factor, with equilibrium respected in every point of the structure during free vibrations.

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- ▶ starting from a shape function $\Psi_0(x)$, a new shape function Ψ_1 can be determined normalizing the displacements due to the inertial forces associated with $\Psi_0(x)$, $f_1 = \bar{m}(x)\Psi_0(x)$,

Rayleigh's Quotient Method

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- we are going to demonstrate that the new shape function is a better approximation of the true mode shape

Selection of mode shapes

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Given different shape functions Ψ_i and considering the true shape of free vibration Ψ , in the former cases equilibrium is not respected by the structure itself.

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To keep inertia induced deformation proportional to Ψ_i we must consider the presence of additional elastic constraints. This leads to the following considerations

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To keep inertia induced deformation proportional to Ψ_i we must consider the presence of additional elastic constraints. This leads to the following considerations

- ▶ the frequency of vibration of a structure with additional constraints is higher than the true natural frequency,
- ▶ the criterium to discriminate between different shape functions is: better shape functions give lower estimates of the natural frequency, the true natural frequency being a lower bound of all estimates.

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In general the selection of trial shapes goes through two steps,

- the analyst considers the flexibilities of different parts of the structure and the presence of symmetries to devise an approximate shape,
- the structure is loaded with constant loads directed as the assumed displacements, the displacements are computed and used as the shape function,
- of course a little practice helps a lot in the the choice of a proper pattern of loading...

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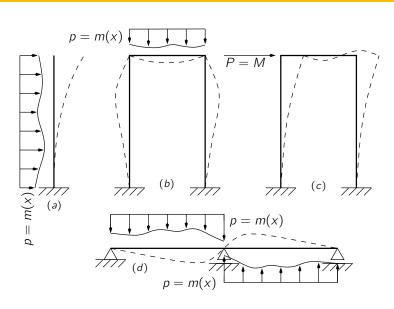
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Choose a trial function $\Psi^{(0)}(x)$ and write

$$v^{(0)} = \Psi^{(0)}(x)Z^{(0)}\sin\omega t$$

$$V_{\text{max}} = \frac{1}{2}Z^{(0)2} \int EJ\Psi^{(0)"2} dx$$

$$T_{\text{max}} = \frac{1}{2}\omega^2 Z^{(0)2} \int \bar{m}\Psi^{(0)2} dx$$

our first estimate R_{00} of ω^2 is

$$\omega^2 = \frac{\int EJ\Psi^{(0)"2} \, dx}{\int \bar{m}\Psi^{(0)2} \, dx}.$$

We try to give a better estimate of
$$V_{\rm max}$$
 computing the external work done by the inertial forces,

$$p^{(0)} = \omega^2 \bar{m}(x) v^{(0)} = Z^{(0)} \omega^2 \Psi^{(0)}(x)$$

the deflections due to $p^{(0)}$ are

$$v^{(1)} = \omega^2 \frac{v^{(1)}}{\omega^2} = \omega^2 \Psi^{(1)} \frac{Z^{(1)}}{\omega^2} = \omega^2 \Psi^{(1)} \bar{Z}^{(1)},$$

where we write $\bar{Z}^{(1)}$ because we need to keep the unknown ω^2 in evidence. The maximum strain energy is

$$V_{\text{max}} = \frac{1}{2} \int p^{(0)} v^{(1)} \, dx = \frac{1}{2} \omega^4 Z^{(0)} \bar{Z}^{(1)} \int \bar{m}(x) \Psi^{(0)} \Psi^{(1)} \, dx$$

Equating to our previous estimate of T_{max} we find the R_{01} estimate

$$\omega^2 = \frac{Z^{(0)}}{\bar{Z}^{(1)}} \frac{\int \bar{m}(x) \Psi^{(0)} \Psi^{(0)} dx}{\int \bar{m}(x) \Psi^{(0)} \Psi^{(1)} dx}$$

Selection of Mode

Refinement of Rayleigh's Estimates

With little additional effort it is possible to compute T_{max} from $v^{(1)}$:

$$T_{\text{max}} = \frac{1}{2}\omega^2 \int \bar{m}(x) v^{(1)2} \, \mathrm{d}x = \frac{1}{2}\omega^6 \bar{Z}^{(1)2} \int \bar{m}(x) \Psi^{(1)2} \, \mathrm{d}x$$

equating to our last approximation for $V_{\rm max}$ we have the R_{11} approximation to the frequency of vibration,

$$\omega^2 = \frac{Z^{(0)}}{\bar{Z}^{(1)}} \frac{\int \bar{m}(x) \Psi^{(0)} \Psi^{(1)} dx}{\int \bar{m}(x) \Psi^{(1)} \Psi^{(1)} dx}.$$

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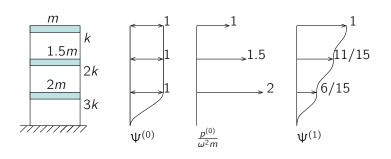
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Refinement Example



$$T = \frac{1}{2}\omega^{2} \times 4.5 \times m Z_{0}^{2}$$

$$V = \frac{1}{2} \times 1 \times 3k Z_{0}^{2}$$

$$V^{(1)} = \frac{15}{4} \frac{m}{k} \omega^{2} \Psi^{(1)}$$

$$V^{(1)} = \frac{1}{2} m \frac{15}{4} \frac{m}{k} \omega^{4} (1 + 33/30 + 4/5)$$

$$= \frac{1}{2} m \frac{15}{4} \frac{m}{k} \omega^{4} \frac{87}{30}$$

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$$\bar{Z}^{(1)} = \frac{15}{4} \frac{m}{k}$$

$$\omega^{2} = \frac{\frac{9}{2} m}{m \frac{87}{8} \frac{m}{k}} = \frac{12}{29} \frac{k}{m} = 0.4138 \frac{k}{m}$$