# SDOF linear oscillator

### Response to Impulsive Loads & Step by Step Methods

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# Nature of Impulsive Loadings

An impulsive load is characterized

- by a single principal impulse, and
- by a relatively short duration.



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# Nature of Impulsive Loadings

An impulsive load is characterized

- by a single principal impulse, and
- by a relatively short duration.



- Impulsive or shock loads are of great importance for the design of certain classes of structural systems, e.g., vehicles or cranes.
- Damping has much less importance in controlling the maximum response to impulsive loadings because the maximum response is reached in a very short time, before the damping forces can dissipate a significant portion of the energy input into the system.
- For this reason, in the following we'll consider only the undamped response to impulsive loads.

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# Definition of Peak Response

When dealing with the response to an impulsive loading of duration  $t_0$  of a SDOF system, with natural period of vibration  $T_n$  we are mostly interested in the *peak response* of the system.

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The **peak response** is the maximum of the absolute value of the response ratio,  $R_{max} = \max{|R(t)|}$ .

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The **peak response** is the maximum of the absolute value of the response ratio,  $R_{max} = \max\{|R(t)|\}$ .

- If t<sub>0</sub> ≪ T<sub>n</sub> necessarily R<sub>max</sub> happens after the end of the loading, and its value can be determined studying the free vibrations of the dynamic system.
- On the other hand, if the excitation lasts *enough* to have at least a local extreme (maximum or minimum) during the excitation we have to consider the more difficult problem of completely determining the response during the application of the impulsive loading.

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## Half-sine Wave Impulse

The sine-wave impulse has expression

$$p(t) = \begin{cases} p_0 \sin \frac{\pi t}{t_0} = p_0 \sin \omega t & \text{for } 0 < t < t_0, \\ 0 & \text{otherwise.} \end{cases}$$

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where  $\omega = \frac{2\pi}{2t_0}$  is the frequency associated with the load.

Note that  $\omega t_0 = \pi$ .

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### Response to sine-wave impulse

Consider an undamped *SDOF* initially at rest, with natural period  $T_n$ , excited by a half-sine impulse of duration  $t_0$ . The frequency ratio is  $\beta = \frac{\tau_n}{2t_0}$  and the response ratio in the interval  $0 < t < t_0$  is

$$R(t) = \frac{1}{1 - \beta^2} \left( \sin \omega t - \beta \sin \frac{\omega t}{\beta} \right),$$
  
$$\dot{R}(t) = \frac{\omega}{1 - \beta^2} \left( \cos \omega t - \cos \frac{\omega t}{\beta} \right). \qquad [\text{NB:} \frac{\omega}{\beta} = \omega_n]$$

It is  $(1 - \beta^2)R(t_0) = -\beta \sin \pi/\beta$  and  $(1 - \beta^2)\dot{R}(t_0) = -\omega (1 + \cos \pi/\beta)$ , consequently for  $t_o \leq t$  the response ratio is

$$R(t) = \frac{-\beta}{1-\beta^2} \left( (1+\cos\frac{\pi}{\beta})\sin\omega_n(t-t_0) + \sin\frac{\pi}{\beta}\cos\omega_n(t-t_0) \right)$$

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### Maximum response to sine impulse

We have an extreme, and a possible peak value, for  $0 \le t \le t_0$  if

$$\dot{R}(t) = \frac{\omega}{1-\beta^2}(\cos \omega t - \cos \frac{\omega t}{\beta}) = 0.$$

That implies that  $\cos \omega t = \cos \omega t / \beta = \cos - \omega t / \beta$ , whose roots are

$$\omega t = \pm \omega t/\beta + 2n\pi, \ n = 0, \pm 1, \pm 2, \pm 3, \ldots$$

It is convenient to substitute  $\omega t = \pi \alpha$ , where  $\alpha = t/t_0$ :

$$\pi a = \pi \left( \mp \frac{a}{\beta} + 2n 
ight), \quad n = 0, \mp 1, \mp 2, \ldots, \quad 0 \leq a \leq 1.$$

Eventually solving for  $\alpha$  one has

$$lpha=rac{2neta}{eta\pm1},\quad n=0,\mp1,\mp2,\ldots,\quad 0$$

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The next slide regards the characteristics of these roots.

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 $\alpha(\beta, n)$ 



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 $\alpha(\beta, n)$ 

In summary, to find the maximum of the response for an assigned  $\beta < 1$ , one has (a) to compute all  $\alpha_k = \frac{2k\beta}{\beta+1}$  until a root is greater than 1, (b) compute all the responses for  $t_k = \alpha_k t_0$ , (c) choose the maximum of the maxima.



- No roots of type  $\alpha_{\min}$  for n > 0;
- no roots of type  $\alpha_{\max}$  for n < 0;
- no roots for  $\beta > 1$ , i.e., no roots for  $t_0 < \frac{T_0}{2}$ ;
- $\begin{array}{l} \text{- only one root of type } \alpha_{\text{max}} \text{ for} \\ \frac{1}{3} < \beta < 1 \text{, i.e., } \frac{T_n}{2} < t_0 < \frac{3T_n}{2} \text{;} \end{array}$
- three roots, two maxima and one minimum, for  $\frac{1}{5} < \beta < \frac{1}{3}$ ;
- five roots, three maxima and two minima, for  $\frac{1}{7}<\beta<\frac{1}{5};$
- etc etc.

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## Maximum response for $\beta > 1$

For  $\beta > 1$ , the maximum response takes place for  $t > t_0$ , and its absolute value (see slide *Response to sine-wave impulse*) is

$$R_{\max} = \frac{\beta}{1-\beta^2} \sqrt{(1+\cos\frac{\pi}{\beta})^2 + \sin^2\frac{\pi}{\beta}},$$

using a simple trigonometric identity we can write

$$R_{\text{max}} = \frac{\beta}{1-\beta^2} \sqrt{2+2\cos\frac{\pi}{\beta}}$$

but  $1+\cos 2\varphi=(\cos^2\varphi+\sin^2\varphi)+(\cos^2\varphi-\sin^2\varphi)=2\cos^2\varphi$  , so that

$$R_{\max} = rac{2eta}{1-eta^2}\cosrac{\pi}{2eta}.$$

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### **Rectangular Impulse**

Consider a rectangular impulse of duration  $t_0$ ,

$$p(t) = p_0 \begin{cases} 1 & \text{for } 0 < t < t_0, \\ 0 & \text{otherwise.} \end{cases} \xrightarrow{p_0} \begin{bmatrix} p_0 & p_0 \\ 0 & p_0 \\ 0 & p_0 \end{bmatrix}$$

The response ratio and its time derivative are

$$R(t) = 1 - \cos \omega_n t, \qquad \dot{R}(t) = \omega_n \sin \omega_n t,$$

and we recognize that we have maxima  $R_{max} = 2$  for  $\omega_n t = n\pi$ , with the condition  $t < t_0$ . Hence we have no maximum during the loading phase for  $t_0 < T_{\rm p}/2$ , and at least one maximum, of value  $2\Delta_{\rm st}$ , if  $t_0 > T_n/2$ .



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## Rectangular Impulse (2)

For shorter impulses, the maximum response ratio is not attained during loading, so we have to compute the amplitude of the free vibrations after the end of loading (remember, as  $t_0 \le T_n/2$  the velocity is positive at  $t = t_0$ !).

$$R(t) = (1 - \cos \omega_n t_0) \cos \omega_n (t - t_0) + (\sin \omega_n t_0) \sin \omega_n (t - t_0).$$

The amplitude of the response ratio is then

$$\begin{aligned} A &= \sqrt{(1 - \cos \omega_{\mathsf{n}} t_0)^2 + \sin^2 \omega_{\mathsf{n}} t_0} = \\ &= \sqrt{2(1 - \cos \omega_{\mathsf{n}} t_0)} = 2 \sin \frac{\omega_{\mathsf{n}} t_0}{2}. \end{aligned}$$

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## **Triangular Impulse**

Let's consider the response of a SDOF to a triangular impulse,



As usual, we must start finding the minimum duration that gives place to a maximum of the response in the loading phase, that is

$$R(t) = rac{1}{\omega_{n}t_{0}} \sin \omega_{n} rac{t}{t_{0}} - \cos \omega_{n} rac{t}{t_{0}} + 1 - rac{t}{t_{0}}, \quad 0 < t < t_{0}.$$

Taking the first derivative and setting it to zero, one can see that the first maximum occurs for  $t = t_0$  for  $t_0 = 0.37101T_n$ , and substituting one can see that  $R_{max} = 1$ .

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For load durations shorter than  $0.37101T_n$ , the maximum occurs after loading and it's necessary to compute the displacement and velocity at the end of the load phase.

For longer loads, the maxima are in the load phase, so that one has to find the all the roots of  $\dot{R}(t)$ , compute all the extreme values and finally sort out the absolute value maximum.

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## Shock or response spectra

We have seen that the response ratio is determined by the ratio of the impulse duration to the natural period of the oscillator. One can plot the maximum displacement ratio  $R_{max}$  as a function of  $t_o/T_n$  for various forms of impulsive loads.



Such plots are commonly known as displacement-response spectra, or simply as response spectra.

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For long duration loadings, the maximum response ratio depends on the rate of the increase of the load to its maximum: for a step function we have a maximum response ratio of 2, for a slowly varying load we tend to a quasi-static response, hence a factor  $\approx 1$ 

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- For long duration loadings, the maximum response ratio depends on the rate of the increase of the load to its maximum: for a step function we have a maximum response ratio of 2, for a slowly varying load we tend to a quasi-static response, hence a factor  $\approx 1$
- On the other hand, for short duration loads, the maximum displacement is in the free vibration phase, and its amplitude depends on the work done on the system by the load.
- The response ratio depends further on the maximum value of the load impulse, so we can say that the maximum displacement is a more significant measure of response.

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An approximate procedure to evaluate the maximum displacement for a short impulse loading is based on the impulse-momentum relationship,

$$m\Delta \dot{x} = \int_0^{t_0} \left[ p(t) - k x(t) \right] dt.$$

When one notes that, for small  $t_0$ , the displacement is of the order of  $t_0^2$  while the velocity is in the order of  $t_0$ , it is apparent that the *kx* term may be dropped from the above expression, i.e.,

$$m\Delta \dot{x} \approx \int_0^{t_0} p(t) dt.$$

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Using the previous approximation, the velocity at time  $t_0$  is

$$\dot{x}(t_0)=\frac{1}{m}\int_0^{t_0}p(t)\ dt,$$

and considering again a negligibly small displacement at the end of the loading,  $x(t_0) \cong 0$ , one has

$$x(t-t_0) \simeq \frac{\int_0^{t_0} p(t) dt}{m\omega_n} \sin \omega_n (t-t_0).$$

Please note that the above equation is exact for an infinitesimal impulse loading.

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Both the Duhamel integral and the Fourier transform methods lie on on the principle of superposition, i.e., superposition of the responses

- to a succession of infinitesimal impulses, using a convolution (Duhamel) integral, when operating in time domain
- to an infinity of infinitesimal harmonic components, using the frequency response function, when operating in frequency domain.

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The principle of superposition implies *linearity*, but this assumption is often invalid, e.g., a severe earthquake is expected to induce inelastic deformation in a code-designed structure.

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The internal state of a linear dynamical system, considering that the mass, the damping and the stiffness do not vary during the excitation, is described in terms of its displacements and its velocity, i.e., the so called *state vector* 

$$\mathbf{x} = \begin{bmatrix} \mathbf{x}(t) \\ \dot{\mathbf{x}}(t) \end{bmatrix}.$$

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$$\mathbf{x} = \begin{bmatrix} \mathbf{x}(t) \\ \dot{\mathbf{x}}(t) \end{bmatrix}.$$

For a non linear system the state vector must include other information, e.g. the current tangent stiffness, the cumulated plastic deformations, the internal damage, ...

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The so-called step-by-step methods restrict the assumption of linearity to the duration of a (usually short) *time step*.

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The so-called step-by-step methods restrict the assumption of linearity to the duration of a (usually short) *time step*.

Given an initial system state, in step-by-step methods we divide the time in *steps* of known, short duration  $h_i$  (usually  $h_i = h$ , a constant) and from the initial system state at the beginning of each step we compute the final system state at the end of each step.

The final state vector in step *i* will be the initial state in the subsequent step, i + 1.

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Operating independently the analysis for each time step there are no requirements for superposition and non linear behaviour can be considered assuming that the structural properties remain constant during each time step.

In many cases, the non linear behaviour can be reasonably approximated by a *local* linear model, valid for the duration of the time step.

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Operating independently the analysis for each time step there are no requirements for superposition and non linear behaviour can be considered assuming that the structural properties remain constant during each time step.

In many cases, the non linear behaviour can be reasonably approximated by a *local* linear model, valid for the duration of the time step.

If the approximation is not good enough, usually a better approximation can be obtained reducing the time step.

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Generality step-by-step methods can deal with every kind of non-linearity, e.g., variation in mass or damping or variation in geometry and not only with mechanical non-linearities.

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Efficiency step-by-step methods are very efficient and are usually preferred also for linear systems in place of Duhamel integral.

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- Efficiency step-by-step methods are very efficient and are usually preferred also for linear systems in place of Duhamel integral.
- Extensibility step-by-step methods can be easily extended to systems with many degrees of freedom, simply using matrices and vectors in place of scalar quantities.

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roundoff using too few digits in calculations.

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roundoff using too few digits in calculations.

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roundoff using too few digits in calculations.

truncation using too few terms in series expressions of quantities, instability the amplification of errors deriving from roundoff, truncation or modeling in one time step in all following time steps, usually depending on the time step duration. SDOF linear oscillator

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Errors may be classified as

phase shifts or change in frequency of the response,

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Errors may be classified as

- phase shifts or change in frequency of the response,
- artificial damping, the numerical procedure removes or adds energy to the dynamic system.

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### Examples of SbS Methods

We use the exact solution of the equation of motion for a system excited by a linearly varying force, so the source of all errors lies in the piecewise linearisation of the force function and in the approximation due to a local linear model.

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Piecewise Exact

- We use the exact solution of the equation of motion for a system excited by a linearly varying force, so the source of all errors lies in the piecewise linearisation of the force function and in the approximation due to a local linear model.
- We will see that an appropriate time step can be decided in terms of the number of points required to accurately describe either the force or the response function.

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Piecewise Exact

# Piecewise exact method

For a generic time step of duration h, consider

- ▶  $\{x_0, \dot{x}_0\}$  the initial state vector,
- *p*<sub>0</sub> and *p*<sub>1</sub>, the values of *p*(*t*) at the start and the end of the integration step,
- the linearised force

$$p(\tau) = p_0 + \alpha \tau, \ 0 \le \tau \le h, \ \alpha = (p(h) - p(0))/h,$$

the forced response

$$\mathbf{x} = e^{-\zeta \omega \tau} (\mathbf{A} \sin(\omega_{\mathsf{D}} \tau) + \mathbf{B} \cos(\omega_{\mathsf{D}} \tau)) + (\alpha k \tau + k p_0 - \alpha c) / k^2,$$

where k and c are the stiffness and damping of the SDOF system.

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#### Piecewise Exact

Evaluating the response *x* and the velocity  $\dot{x}$  for  $\tau = 0$  and equating to  $\{x_0, \dot{x}_0\}$ , writing  $\Delta_{st} = p(0)/k$  and  $\delta(\Delta_{st}) = (p(h) - p(0))/k$ , one can find *A* and *B* 

$$A = \left(\dot{x}_0 + \zeta \omega B - \frac{\delta(\Delta_{st})}{h}\right) \frac{1}{\omega_0}$$
$$B = x_0 + \frac{2\zeta}{\omega} \frac{\delta(\Delta_{st})}{h} - \Delta_{st}$$

substituting and evaluating for  $\tau = h$  one finds the state vector at the end of the step.

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### Piecewise exact method

With

$$\mathcal{S}_{\zeta,h} = \sin(\omega_{\mathsf{D}}h) \exp(-\zeta \omega h)$$
 and  $\mathcal{C}_{\zeta,h} = \cos(\omega_{\mathsf{D}}h) \exp(-\zeta \omega h)$ 

and the previous definitions of  $\Delta_{st}$  and  $\delta(\Delta_{st})$ , finally we can write

$$\begin{aligned} \mathbf{x}(h) &= A \, \mathcal{S}_{\zeta,h} + B \, \mathcal{C}_{\zeta,h} + (\Delta_{st} + \delta(\Delta_{st})) - \frac{2\zeta}{\omega} \frac{\delta(\Delta_{st})}{h} \\ \dot{\mathbf{x}}(h) &= A(\omega_{\mathsf{D}} \mathcal{C}_{\zeta,h} - \zeta \omega \mathcal{S}_{\zeta,h}) - B(\zeta \omega \mathcal{C}_{\zeta,h} + \omega_{\mathsf{D}} \mathcal{S}_{\zeta,h}) + \frac{\delta(\Delta_{st})}{h} \end{aligned}$$

where

$$B = x_0 + \frac{2\zeta}{\omega} \frac{\delta(\Delta_{st})}{h} - \Delta_{st}, \quad A = \left(\dot{x}_0 + \zeta \omega B - \frac{\delta(\Delta_{st})}{h}\right) \frac{1}{\omega_D}.$$

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# Example

We have a damped system that is excited by a load in resonance with the system, we know the exact response

$$x(t) = \Delta_{\rm st} \big( (\frac{\sin \omega_D t}{2\sqrt{1-\zeta^2}} + \frac{\cos \omega_D t}{2\zeta}) \exp(-\zeta \omega_n t) - \frac{\cos \omega_n t}{2\zeta} \big)$$

and we want to compute a step-by-step approximation using different step lengths.

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# Example

We have a damped system that is excited by a load in resonance with the system, we know the exact response

$$x(t) = \Delta_{\rm st} \left( \left( \frac{\sin \omega_D t}{2\sqrt{1-\zeta^2}} + \frac{\cos \omega_D t}{2\zeta} \right) \exp(-\zeta \omega_n t) - \frac{\cos \omega_n t}{2\zeta} \right)$$

and we want to compute a step-by-step approximation using different step lengths.



It is apparent that you have a very good approximation when the linearised loading is a very good approximation of the input function, let's say  $h \leq T/10$ .

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# **Central differences**

To derive the Central Differences Method, we write the eq. of motion at time  $\tau=0$  and find the initial acceleration,

$$m\ddot{x}_0 + c\dot{x}_0 + kx_0 = p_0 \Rightarrow \ddot{x}_0 = \frac{1}{m}(p_0 - c\dot{x}_0 - kx_0)$$

On the other hand, the initial acceleration can be expressed in terms of finite differences,

$$\ddot{x}_0 = \frac{x_1 - 2x_0 + x_{-1}}{h^2} = \frac{1}{m}(p_0 - c\dot{x}_0 - kx_0)$$

solving for  $x_1$ 

$$x_1 = 2x_0 - x_{-1} + \frac{h^2}{m}(p_0 - c\dot{x}_0 - kx_0)$$

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# **Central differences**

We have an expression for  $x_1$ , the displacement at the end of the step,

$$x_1 = 2x_0 - x_{-1} + \frac{h^2}{m}(p_0 - c\dot{x}_0 - kx_0),$$

but we have an additional unknown,  $x_{-1}$ ... if we write the finite differences approximation to  $\dot{x}_0$  we can find an approximation to  $x_{-1}$  in terms of the initial velocity  $\dot{x}_0$  and the unknown  $x_1$ 

$$\dot{x}_0 = rac{x_1 - x_{-1}}{2h} \Rightarrow x_{-1} = x_1 - 2h\dot{x}_0$$

Substituting in the previous equation

$$x_1 = 2x_0 - x_1 + 2h\dot{x}_0 + \frac{h^2}{m}(p_0 - c\dot{x}_0 - kx_0),$$

and solving for  $x_1$ 

$$x_1 = x_0 + h\dot{x}_0 + \frac{h^2}{2m}(p_0 - c\dot{x}_0 - kx_0)$$

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# **Central differences**

$$x_1 = x_0 + h\dot{x}_0 + \frac{h^2}{2m}(p_0 - c\dot{x}_0 - kx_0)$$

To start a new step, we need the value of  $\dot{x}_1$ , but we may approximate the mean velocity, again, by finite differences

$$\frac{\dot{x}_0 + \dot{x}_1}{2} = \frac{x_1 - x_0}{h} \Rightarrow \dot{x}_1 = \frac{2(x_1 - x_0)}{h} - \dot{x}_0$$

The method is very simple, but it is *conditionally stable*. The stability condition is defined with respect to the natural frequency, or the natural period, of the SDOF oscillator,

$$\omega_{n}h \leq 2 \Rightarrow h \leq \frac{T_{n}}{\pi} \approx 0.32T_{n}$$

For a SDOF this is not relevant because, as we have seen in our previous example, we need more points for response cycle to correctly represent the response.

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Examples of SbS Methods Piecewise Exact Central Differences Integration Constant Acceleration Linear Acceleration Newmark Beta Non Linear Systems Newton-Raphson We will make use of an *hypothesis* on the variation of the acceleration during the time step and of analytical integration of acceleration and velocity to step forward from the initial to the final condition for each time step.

In general, these methods are based on the two equations

$$\dot{x}_1 = \dot{x}_0 + \int_0^h \ddot{x}(\tau) d\tau,$$
$$x_1 = x_0 + \int_0^h \dot{x}(\tau) d\tau,$$

which express the final velocity and the final displacement in terms of the initial values  $x_0$  and  $\dot{x}_0$  and some definite integrals that depend on the *assumed* variation of the acceleration during the time step.

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the constant acceleration method,

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- the constant acceleration method,
- the linear acceleration method,

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Central Difference:

Integration

- the constant acceleration method,
- the linear acceleration method,
- the family of methods known as Newmark Beta Methods, that comprises the previous methods as particular cases.

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### **Constant Acceleration**

Here we assume that the acceleration is constant during each time step, equal to the mean value of the initial and final values:

$$\ddot{\mathbf{x}}(\tau) = \ddot{\mathbf{x}}_0 + \Delta \ddot{\mathbf{x}}/2,$$

where  $\Delta \ddot{x} = \ddot{x}_1 - \ddot{x}_0$ , hence

$$\dot{x}_1 = \dot{x}_0 + \int_0^h (\ddot{x}_0 + \Delta \ddot{x}/2) d\tau$$
$$\Rightarrow \Delta \dot{x} = \ddot{x}_0 h + \Delta \ddot{x} h/2$$
$$x_1 = x_0 + \int_0^h (\dot{x}_0 + (\ddot{x}_0 + \Delta \ddot{x}/2)\tau) d\tau$$
$$\Rightarrow \Delta x = \dot{x}_0 h + (\ddot{x}_0) h^2/2 + \Delta \ddot{x} h^2/4$$

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Taking into account the two equations on the right of the previous slide, and solving for  $\Delta \dot{x}$  and  $\Delta \ddot{x}$  in terms of  $\Delta x$ , we have

$$\Delta \dot{x} = rac{2\Delta x - 2h\dot{x}_0}{h}, \quad \Delta \ddot{x} = rac{4\Delta x - 4h\dot{x}_0 - 2\ddot{x}_0h^2}{h^2}.$$

We have two equations and three unknowns... Assuming that the system characteristics are constant during a single step, we can write the equation of motion at times  $\tau = h$  and  $\tau = 0$ , subtract member by member and write the *incremental equation of motion* 

 $m\Delta\ddot{x} + c\Delta\dot{x} + k\Delta x = \Delta p$ ,

that is a third equation that relates our unknowns.

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### **Constant acceleration**

Substituting the above expressions for  $\Delta \dot{x}$  and  $\Delta \ddot{x}$  in the incremental eq. of motion and solving for  $\Delta x$  gives, finally,

$$\Delta x = \frac{\tilde{p}}{\tilde{k}}, \qquad \Delta \dot{x} = \frac{2\Delta x - 2h\dot{x}_0}{h}$$

where

$$\tilde{k} = k + \frac{2c}{h} + \frac{4m}{h^2}$$
$$\tilde{p} = \Delta p + 2c\dot{x}_0 + m(2\ddot{x}_0 + \frac{4}{h}\dot{x}_0)$$

While it is possible to compute the final acceleration in terms of  $\Delta x$ , to achieve a better accuracy it is usually computed solving the equation of equilibrium written at the end of the time step.

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Two further remarks

- 1. The method is unconditionally stable
- 2. The effective stiffness, disregarding damping, is  $\tilde{k} \approx k + 4m/h^2$ .

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- 1. The method is unconditionally stable
- 2. The effective stiffness, disregarding damping, is  $\tilde{k} \approx k + 4m/h^2$ .

Dividing both members of the above equation by k it is

$$rac{ ilde{k}}{k} = 1 + rac{4}{\omega_{\mathrm{n}}^2 h^2} = 1 + rac{4}{(2\pi/T_{\mathrm{n}})^2 h^2} = rac{T_{\mathrm{n}}^2}{\pi^2 h^2},$$

The number  $n_T$  of time steps in a period  $T_n$  is related to the time step duration,  $n_T = T_n/h$ , solving for h and substituting in our last equation, we have

$$rac{ ilde{k}}{k}pprox \mathbf{1}+rac{n_{ extsf{T}}^2}{\pi^2}$$

For, e.g.,  $n_T = 2\pi$  it is  $\tilde{k}/k \approx 1 + 4$ , the mass contribution to the effective stiffness is four times the elastic stiffness and the 80% of the total.

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### **Linear Acceleration**

We assume that the acceleration is linear, i.e.

$$\ddot{x}(t) = \ddot{x}_0 + \Delta \ddot{x} \frac{\tau}{h}$$

hence

$$\Delta \dot{x} = \ddot{x}_0 h + \Delta \ddot{x} h/2, \quad \Delta x = \dot{x}_0 h + \ddot{x}_0 h^2/2 + \Delta \ddot{x} h^2/6$$

Following a derivation similar to what we have seen in the case of constant acceleration, we can write, again,

$$\Delta x = \left(k + 3\frac{c}{h} + 6\frac{m}{h^2}\right)^{-1} \left[\Delta p + c(\ddot{x}_0\frac{h}{2} + 3\dot{x}_0) + m(3\ddot{x}_0 + 6\frac{\dot{x}_0}{h})\right]$$
$$\Delta \dot{x} = \Delta x\frac{3}{h} - 3\dot{x}_0 - \ddot{x}_0\frac{h}{2}$$

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Examples of SbS Methods Piecewise Exact Central Differences Integration Constant Acceleration Linear Acceleration Newmark Beta Non Linear Systems Newtron Bankson The linear acceleration method is *conditionally stable*, the stability condition being

$$rac{h}{T} \leq rac{\sqrt{3}}{\pi} pprox 0.55$$

When dealing with SDOF systems, this condition is never of concern, as we need a shorter step to accurately describe the response of the oscillator, let's say  $h \le 0.12T$ ...

When stability is not a concern, the accuracy of the linear acceleration method is far superior to the accuracy of the constant acceleration method, so that this is the method of choice for the analysis of SDOF systems.

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$$\begin{aligned} \Delta \dot{x} &= (1 - \gamma)h\ddot{x}_0 + \gamma h\ddot{x}_1\\ \Delta x &= h\dot{x}_0 + (\frac{1}{2} - \beta)h^2\ddot{x}_0 + \beta h^2\ddot{x}_1 \end{aligned}$$

The factor  $\gamma$  weights the influence of the initial and final accelerations on the velocity increment, while  $\beta$  has a similar role with respect to the displacement increment.

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Examples of SbS Methods Piecewise Exact Central Differences Integration Constant Acceleration Linear Acceleration Newmark Beta Non Linear Systems Newton-Raphson Using  $\gamma \neq 1/2$  leads to numerical damping, so when analysing SDOF systems, one uses  $\gamma = 1/2$  (numerical damping may be desirable when dealing with MDOF systems).

Using  $\beta = \frac{1}{4}$  leads to the constant acceleration method, while  $\beta = \frac{1}{6}$  leads to the linear acceleration method. In the context of MDOF analysis, it's worth knowing what is the minimum  $\beta$  that leads to an unconditionally stable behaviour.

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# Newmark Beta Methods

The general format for the solution of the incremental equation of motion using the Newmark Beta Method can be written as follows:

$$\Delta x = \frac{\Delta \tilde{\rho}}{\tilde{k}}$$
$$\Delta v = \frac{\gamma}{\beta} \frac{\Delta x}{h} - \frac{\gamma}{\beta} v_0 + h \left(1 - \frac{\gamma}{2\beta}\right) a_0$$

with

$$\tilde{k} = k + \frac{\gamma}{\beta}\frac{c}{h} + \frac{1}{\beta}\frac{m}{h^2}$$
$$\Delta \tilde{p} = \Delta p + \left(h\left(\frac{\gamma}{2\beta} - 1\right)c + \frac{1}{2\beta}m\right)a_0 + \left(\frac{\gamma}{\beta}c + \frac{1}{\beta}\frac{m}{h}\right)v_0$$

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Non Linear Systems Newton-Raphson A convenient procedure for integrating the response of a non linear system is based on the incremental formulation of the equation of motion, where for the stiffness and the damping were taken values representative of their variation during the time step: in line of principle, the mean values of stiffness and damping during the time step, or, as this is usually not possible, their initial values,  $k_0$  and  $c_0$ . The Newton-Raphson method can be used to reduce the unbalanced forces at the end of the step.

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# Non Linear Systems

Usually we use the modified Newton-Raphson method, characterised by not updating the system stiffness at each iteration. In pseudo-code, referring for example to the Newmark Beta Method

```
x1,v1,f1 = x0,v0,f0 % initialisation; gb=gamma/beta
Dr = DpTilde
loop:
   Dx = Dr/kTilde
   x^{2} = x^{1} + Dx
   v^{2} = gb^{*}Dx/h + (1-gb)^{*}v^{1} + (1-gb/2)^{*}h^{*}a^{0}
   x pl = update u pl(...)
   f_2 = k^*(x_2 - x p_1)
   % important
   Df = (f2-f1) + (kTilde-k ini)*Dx
   Dr = Dr - Df
   x1. v1, f1 = x2, v2, f2
   if (tol(...) < reg tol) BREAK loop
```

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Use the constant acceleration method to integrate the response, with h = 0.05s and, successively, h = 0.02s. Note that the stiffness is either 0 or k, write down the expression for the effective stiffness and loading in the incremental formulation, write a spreadsheet or a program to make the computations.

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