

# SDOF linear oscillator

## Response to Impulsive Loads & Step by Step Methods

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# Outline

Response to Impulsive Loading

Review of Numerical Methods

Step-by-step Methods

Examples of SbS Methods

- Constant Acceleration Method

- Linear Acceleration Method

- Newmark Beta Methods

- Modified Newton-Raphson Method

## **Impulsive Loads**

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# Response to Impulsive Loadings

## Response to Impulsive Loading

Introduction

Response to Half-Sine Wave Impulse

Response for Rectangular and Triangular Impulses

Shock or response spectra

Approximate Analysis of Response Peak

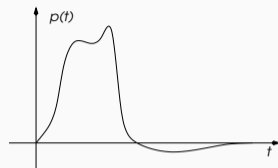
Review of Numerical Methods

Step-by-step Methods

## Nature of Impulsive Loadings

An impulsive load is characterized

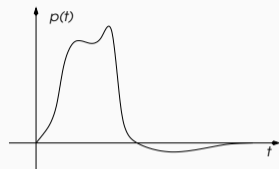
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- Impulsive or shock loads are of great importance for the design of certain classes of structural systems, e.g., vehicles or cranes.
- Damping has much less importance in controlling the maximum response to impulsive loadings because the maximum response is reached in a very short time, before the damping forces can dissipate a significant portion of the energy input into the system.
- For this reason, in the following we'll consider only the undamped response to impulsive loads.

## Definition of *Peak Response*

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- If  $t_0 \ll T_n$  necessarily  $R_{max}$  happens after the end of the loading, and its value can be determined studying the free vibrations of the dynamic system.
- On the other hand, if the excitation lasts *enough* to have at least a local extreme (maximum or minimum) during the excitation we have to consider the more difficult problem of completely determining the response during the application of the impulsive loading.

## Half-sine Wave Impulse

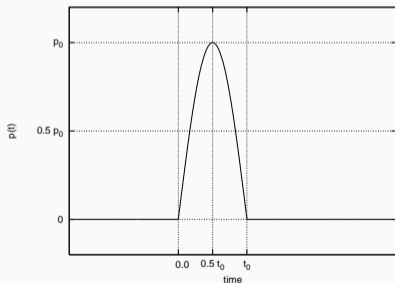
The sine-wave impulse has expression

$$p(t) = \begin{cases} p_0 \sin \frac{\pi t}{t_0} = p_0 \sin \omega t & \text{for } 0 < t < t_0, \\ 0 & \text{otherwise.} \end{cases}$$

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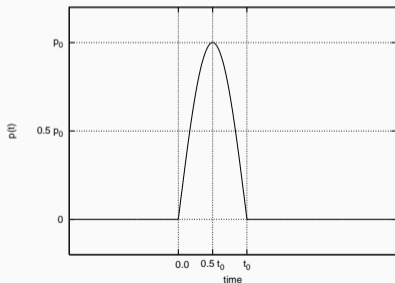
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where  $\omega = \frac{2\pi}{2t_0}$  is the frequency associated with the load.

Note that  $\omega t_0 = \pi$ .

## Response to sine-wave impulse

Consider an undamped *SDOF* initially at rest, with natural period  $T_n$ , excited by a half-sine impulse of duration  $t_0$ .

The frequency ratio is  $\beta = T_n/2t_0$  and the response ratio in the interval  $0 < t < t_0$  is

$$R(t) = \frac{1}{1 - \beta^2} \left( \sin \omega t - \beta \sin \frac{\omega t}{\beta} \right),$$

$$\dot{R}(t) = \frac{\omega}{1 - \beta^2} \left( \cos \omega t - \cos \frac{\omega t}{\beta} \right). \quad [\text{NB: } \frac{\omega}{\beta} = \omega_n]$$

It is  $(1 - \beta^2)R(t_0) = -\beta \sin \pi/\beta$  and  $(1 - \beta^2)\dot{R}(t_0) = -\omega (1 + \cos \pi/\beta)$ , consequently for  $t_0 \leq t$  the response ratio is

$$R(t) = \frac{-\beta}{1 - \beta^2} \left( \left(1 + \cos \frac{\pi}{\beta}\right) \sin \omega_n (t - t_0) + \sin \frac{\pi}{\beta} \cos \omega_n (t - t_0) \right)$$

## Maximum response to sine impulse

We have an extreme, and a possible peak value, for  $0 \leq t \leq t_0$  if

$$\dot{R}(t) = \frac{\omega}{1 - \beta^2} (\cos \omega t - \cos \frac{\omega t}{\beta}) = 0.$$

That implies that  $\cos \omega t = \cos \omega t / \beta = \cos -\omega t / \beta$ , whose roots are

$$\omega t = \mp \omega t / \beta + 2n\pi, \quad n = 0, \mp 1, \mp 2, \mp 3, \dots$$

It is convenient to substitute  $\omega t = \pi\alpha$ , where  $\alpha = t/t_0$ :

$$\pi\alpha = \pi \left( \mp \frac{\alpha}{\beta} + 2n \right), \quad n = 0, \mp 1, \mp 2, \dots, \quad 0 \leq \alpha \leq 1.$$

Eventually solving for  $\alpha$  one has

$$\alpha = \frac{2n\beta}{\beta \pm 1}, \quad n = 0, \mp 1, \mp 2, \dots, \quad 0 < \alpha < 1.$$

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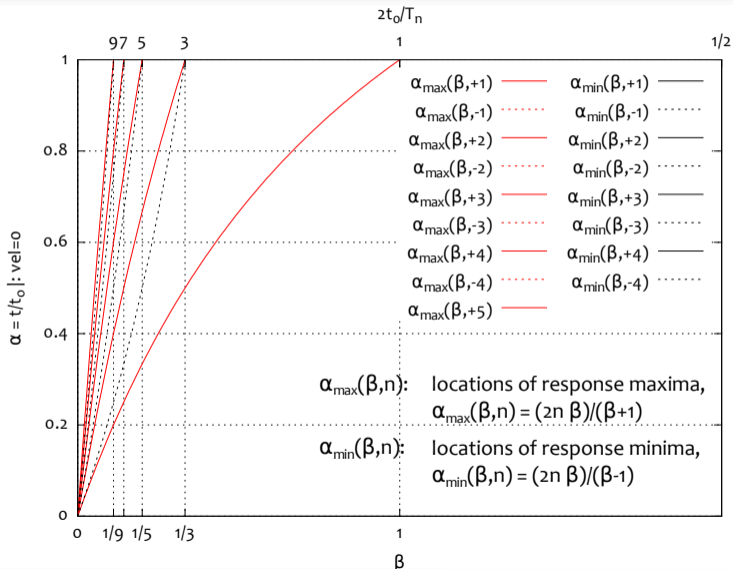
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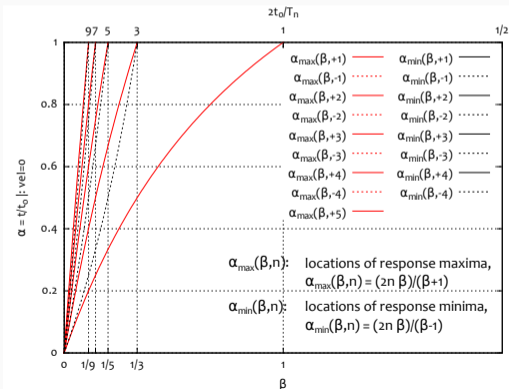
The next slide regards the characteristics of these roots.

$\alpha(\beta, n)$ 



# $\alpha(\beta, n)$

In summary, to find the maximum of the response for an assigned  $\beta < 1$ , one has (a) to compute all  $\alpha_k = \frac{2k\beta}{\beta+1}$  until a root is greater than 1, (b) compute all the responses for  $t_k = \alpha_k t_0$ , (c) choose the maximum of the maxima.



- No roots of type  $\alpha_{\min}$  for  $n > 0$ ;
- no roots of type  $\alpha_{\max}$  for  $n < 0$ ;
- no roots for  $\beta > 1$ , i.e., no roots for  $t_0 < \frac{T_n}{2}$ ;
- only one root of type  $\alpha_{\max}$  for  $\frac{1}{3} < \beta < 1$ , i.e.,  $\frac{T_n}{2} < t_0 < \frac{3T_n}{2}$ ;
- three roots, two maxima and one minimum, for  $\frac{1}{5} < \beta < \frac{1}{3}$ ;
- five roots, three maxima and two minima, for  $\frac{1}{7} < \beta < \frac{1}{5}$ ;

## Maximum response for $\beta > 1$

For  $\beta > 1$ , the maximum response takes place for  $t > t_0$ , and its absolute value (see slide *Response to sine-wave impulse*) is

$$R_{\max} = \frac{\beta}{1 - \beta^2} \sqrt{(1 + \cos \frac{\pi}{\beta})^2 + \sin^2 \frac{\pi}{\beta}},$$

using a simple trigonometric identity we can write

$$R_{\max} = \frac{\beta}{1 - \beta^2} \sqrt{2 + 2 \cos \frac{\pi}{\beta}}$$

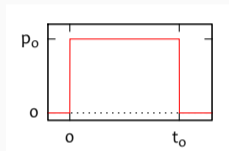
but  $1 + \cos 2\phi = (\cos^2 \phi + \sin^2 \phi) + (\cos^2 \phi - \sin^2 \phi) = 2 \cos^2 \phi$ , so that

$$R_{\max} = \frac{2\beta}{1 - \beta^2} \cos \frac{\pi}{2\beta}.$$

## Rectangular Impulse

Consider a rectangular impulse of duration  $t_0$ ,

$$p(t) = p_0 \begin{cases} 1 & \text{for } 0 < t < t_0, \\ 0 & \text{otherwise.} \end{cases}$$



The response ratio and its time derivative are

$$R(t) = 1 - \cos \omega_n t, \quad \dot{R}(t) = \omega_n \sin \omega_n t,$$

and we recognize that we have maxima  $R_{\max} = 2$  for  $\omega_n t = n\pi$ , with the condition  $t \leq t_0$ . Hence we have no maximum during the loading phase for  $t_0 < T_n/2$ , and at least one maximum, of value  $2\Delta_{st}$ , if  $t_0 \geq T_n/2$ .

## Rectangular Impulse (2)

For shorter impulses, the maximum response ratio is not attained during loading, so we have to compute the amplitude of the free vibrations after the end of loading (remember, as  $t_0 \leq T_n/2$  the velocity is positive at  $t = t_0$ !).

$$R(t) = (1 - \cos \omega_n t_0) \cos \omega_n (t - t_0) + (\sin \omega_n t_0) \sin \omega_n (t - t_0).$$

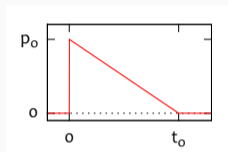
The amplitude of the response ratio is then

$$\begin{aligned} A &= \sqrt{(1 - \cos \omega_n t_0)^2 + \sin^2 \omega_n t_0} = \\ &= \sqrt{2(1 - \cos \omega_n t_0)} = 2 \sin \frac{\omega_n t_0}{2}. \end{aligned}$$

## Triangular Impulse

Let's consider the response of a *SDOF* to a triangular impulse,

$$p(t) = p_0 (1 - t/t_0) \text{ for } 0 < t < t_0$$



As usual, we must start finding the minimum duration that gives place to a maximum of the response in the loading phase, that is

$$R(t) = \frac{1}{\omega_n t_0} \sin \omega_n \frac{t}{t_0} - \cos \omega_n \frac{t}{t_0} + 1 - \frac{t}{t_0}, \quad 0 < t < t_0.$$

Taking the first derivative and setting it to zero, one can see that the first maximum occurs for  $t = t_0$  for  $t_0 = 0.37101T_n$ , and substituting one can see that  $R_{\max} = 1$ .

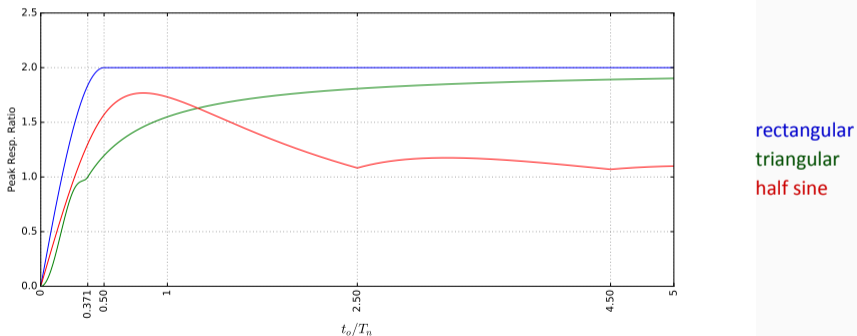
## Triangular Impulse (2)

For load durations shorter than  $0.37101T_n$ , the maximum occurs after loading and it's necessary to compute the displacement and velocity at the end of the load phase.

For longer loads, the maxima are in the load phase, so that one has to find the all the roots of  $\dot{R}(t)$ , compute all the extreme values and finally sort out the absolute value maximum.

## Shock or response spectra

We have seen that the response ratio is determined by the ratio of the impulse duration to the natural period of the oscillator. One can plot the maximum displacement ratio  $R_{\max}$  as a function of  $t_o/T_n$  for various forms of impulsive loads.



Such plots are commonly known as displacement-response spectra, or simply as response spectra.

## Approximate Analysis

For long duration loadings, the maximum response ratio depends on the rate of the increase of the load to its maximum: for a step function we have a maximum response ratio of 2, for a slowly varying load we tend to a quasi-static response, hence a factor  $\cong 1$



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On the other hand, for short duration loads, the maximum displacement is in the free vibration phase, and its amplitude depends on the work done on the system by the load.

The response ratio depends further on the maximum value of the load impulse, so we can say that the maximum displacement is a more significant measure of response.

## Approximate Analysis (2)

An approximate procedure to evaluate the maximum displacement for a short impulse loading is based on the impulse-momentum relationship,

$$m\Delta\dot{x} = \int_0^{t_0} [p(t) - kx(t)] dt.$$

When one notes that, for small  $t_0$ , the displacement is of the order of  $t_0^2$  while the velocity is in the order of  $t_0$ , it is apparent that the  $kx$  term may be dropped from the above expression, i.e.,

$$m\Delta\dot{x} \cong \int_0^{t_0} p(t) dt.$$

## Approximate Analysis (3)

Using the previous approximation, the velocity at time  $t_0$  is

$$\dot{x}(t_0) = \frac{1}{m} \int_0^{t_0} p(t) dt,$$

and considering again a negligibly small displacement at the end of the loading,  $x(t_0) \cong 0$ , one has

$$x(t - t_0) \cong \frac{\int_0^{t_0} p(t) dt}{m\omega_n} \sin \omega_n(t - t_0).$$

Please note that the above equation is exact for an infinitesimal impulse loading.

# Review

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# Review of Numerical Methods

Response to Impulsive Loading

Review of Numerical Methods

Linear Methods in Time and Frequency Domain

Step-by-step Methods

Examples of SbS Methods

Constant Acceleration Method

Linear Acceleration Method

Newmark Beta Methods

Modified Newton-Raphson Method

## Previous Methods

Both the Duhamel integral and the Fourier transform methods lie on on the principle of superposition, i.e., superposition of the responses

- to a succession of infinitesimal impulses, using a convolution (Duhamel) integral, when operating in time domain
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The principle of superposition implies *linearity*, but this assumption is often invalid, e.g., **a severe earthquake is expected to induce inelastic deformation in a code-designed structure.**

## State Vector, Linear and Non Linear Systems

The internal state of a linear dynamical system, considering that the mass, the damping and the stiffness do not vary during the excitation, is described in terms of its displacements and its velocity, i.e., the so called *state vector*

$$x = \begin{bmatrix} x(t) \\ \dot{x}(t) \end{bmatrix}.$$



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For a non linear system the state vector must include other information, e.g. the current tangent stiffness, the cumulated plastic deformations, the internal damage, ...

## **SbS Methods**

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# Step-By-Step Methods

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Step-by-step Methods

Introduction to Step-by-step Methods

Criticism of SbS Methods

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Given an initial system state, in step-by-step methods we divide the time in *steps* of known, short duration  $h_i$  (usually  $h_i = h$ , a constant) and from the initial system state at the beginning of each step we compute the final system state at the end of each step.

The final state vector in step  $i$  will be the initial state in the subsequent step,  $i + 1$ .

## Step-by-step Methods, 2

Operating independently the analysis for each time step there are no requirements for superposition and non linear behaviour can be considered assuming that the structural properties remain constant during each time step.

In many cases, the non linear behaviour can be reasonably approximated by a *local* linear model, valid for the duration of the time step.

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In many cases, the non linear behaviour can be reasonably approximated by a *local* linear model, valid for the duration of the time step.

If the approximation is not good enough, usually a better approximation can be obtained reducing the time step.

## Advantages of s-b-s methods

**Generality** step-by-step methods can deal with every kind of non-linearity, e.g., variation in mass or damping or variation in geometry and not only with mechanical non-linearities.



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- Efficiency** step-by-step methods are very efficient and are usually preferred also for linear systems in place of Duhamel integral.
- Extensibility** step-by-step methods can be easily extended to systems with many degrees of freedom, simply using matrices and vectors in place of scalar quantities.

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Errors may be classified as

- phase shifts or change in frequency of the response,
- **artificial damping, the numerical procedure removes or adds energy to the dynamic system.**

## **SbS Examples**

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## Examples Of SbS Methods

Response to Impulsive Loading

Review of Numerical Methods

Step-by-step Methods

Examples of SbS Methods

Piecewise Exact Method

Central Differences Method

Methods based on Integration

Constant Acceleration Method

Linear Acceleration Method

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- We will see that an appropriate time step can be decided in terms of the number of points required to accurately describe either the force or the response function.

## Piecewise exact method

For a generic time step of duration  $h$ , consider

- $\{x_0, \dot{x}_0\}$  the initial state vector,
- $p_0$  and  $p_1$ , the values of  $p(t)$  at the start and the end of the integration step,
- the linearised force

$$p(\tau) = p_0 + \alpha\tau, \quad 0 \leq \tau \leq h, \quad \alpha = (p(h) - p(0))/h,$$

- the forced response

$$x = e^{-\zeta\omega\tau}(A \sin(\omega_D\tau) + B \cos(\omega_D\tau)) + (\alpha k\tau + kp_0 - \alpha c)/k^2,$$

where  $k$  and  $c$  are the stiffness and damping of the SDOF system.

## Piecewise exact method

Evaluating the response  $x$  and the velocity  $\dot{x}$  for  $\tau = 0$  and equating to  $\{x_0, \dot{x}_0\}$ , writing  $\Delta_{st} = p(0)/k$  and  $\delta(\Delta_{st}) = (p(h) - p(0))/k$ , one can find  $A$  and  $B$

$$A = \left( \dot{x}_0 + \zeta \omega B - \frac{\delta(\Delta_{st})}{h} \right) \frac{1}{\omega_D}$$
$$B = x_0 + \frac{2\zeta}{\omega} \frac{\delta(\Delta_{st})}{h} - \Delta_{st}$$

substituting and evaluating for  $\tau = h$  one finds the state vector at the end of the step.

## Piecewise exact method

With

$$S_{\zeta,h} = \sin(\omega_D h) \exp(-\zeta \omega h) \text{ and } C_{\zeta,h} = \cos(\omega_D h) \exp(-\zeta \omega h)$$

and the previous definitions of  $\Delta_{st}$  and  $\delta(\Delta_{st})$ , finally we can write

$$x(h) = A S_{\zeta,h} + B C_{\zeta,h} + (\Delta_{st} + \delta(\Delta_{st})) - \frac{2\zeta}{\omega} \frac{\delta(\Delta_{st})}{h}$$

$$\dot{x}(h) = A(\omega_D C_{\zeta,h} - \zeta \omega S_{\zeta,h}) - B(\zeta \omega C_{\zeta,h} + \omega_D S_{\zeta,h}) + \frac{\delta(\Delta_{st})}{h}$$

where

$$B = x_0 + \frac{2\zeta}{\omega} \frac{\delta(\Delta_{st})}{h} - \Delta_{st}, \quad A = \left( \dot{x}_0 + \zeta \omega B - \frac{\delta(\Delta_{st})}{h} \right) \frac{1}{\omega_D}.$$



## Example

We have a damped system that is excited by a load in resonance with the system, we know the exact response

$$x(t) = \Delta_{st} \left( \left( \frac{\sin \omega_D t}{2\sqrt{1-\zeta^2}} + \frac{\cos \omega_D t}{2\zeta} \right) \exp(-\zeta \omega_n t) - \frac{\cos \omega_n t}{2\zeta} \right)$$

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$$m=1000\text{kg,}$$

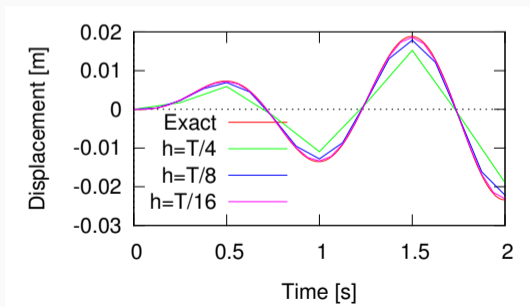
$$k=4\pi^2 \text{ 1000N/m,}$$

$$\omega=2\pi,$$

$$\zeta=0.05,$$

$$p(t) =$$

$$4\pi^2 5 \text{ N } \sin(2\pi t)$$



It is apparent that you have a very good approximation when the linearised

## Central differences

To derive the Central Differences Method, we write the eq. of motion at time  $\tau = 0$  and find the initial acceleration,

$$m\ddot{x}_0 + c\dot{x}_0 + kx_0 = p_0 \Rightarrow \ddot{x}_0 = \frac{1}{m}(p_0 - c\dot{x}_0 - kx_0)$$

On the other hand, the initial acceleration can be expressed in terms of finite differences,

$$\ddot{x}_0 = \frac{x_1 - 2x_0 + x_{-1}}{h^2} = \frac{1}{m}(p_0 - c\dot{x}_0 - kx_0)$$

solving for  $x_1$

$$x_1 = 2x_0 - x_{-1} + \frac{h^2}{m}(p_0 - c\dot{x}_0 - kx_0)$$

## Central differences

We have an expression for  $x_1$ , the displacement at the end of the step,

$$x_1 = 2x_0 - x_{-1} + \frac{h^2}{m}(p_0 - c\dot{x}_0 - kx_0),$$

but we have an additional unknown,  $x_{-1}$ ... if we write the finite differences approximation to  $\dot{x}_0$  we can find an approximation to  $x_{-1}$  in terms of the initial velocity  $\dot{x}_0$  and the unknown  $x_1$

$$\dot{x}_0 = \frac{x_1 - x_{-1}}{2h} \Rightarrow x_{-1} = x_1 - 2h\dot{x}_0$$

Substituting in the previous equation

$$x_1 = 2x_0 - x_1 + 2h\dot{x}_0 + \frac{h^2}{m}(p_0 - c\dot{x}_0 - kx_0),$$

and solving for  $x_1$

$$x_1 = x_0 + h\dot{x}_0 + \frac{h^2}{2m}(p_0 - c\dot{x}_0 - kx_0)$$

## Central differences

$$x_1 = x_0 + h\dot{x}_0 + \frac{h^2}{2m}(p_0 - c\dot{x}_0 - kx_0)$$

To start a new step, we need the value of  $\dot{x}_1$ , but we may approximate the mean velocity, again, by finite differences

$$\frac{\dot{x}_0 + \dot{x}_1}{2} = \frac{x_1 - x_0}{h} \Rightarrow \dot{x}_1 = \frac{2(x_1 - x_0)}{h} - \dot{x}_0$$

The method is very simple, but it is *conditionally stable*. The stability condition is defined with respect to the natural frequency, or the natural period, of the SDOF oscillator,

$$\omega_n h \leq 2 \Rightarrow h \leq \frac{T_n}{\pi} \approx 0.32T_n$$

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*For a SDOF this is not relevant because, as we have seen in our previous example, we need more points for response cycle to correctly represent the response.*

## Methods based on Integration

We will make use of an *hypothesis* on the variation of the acceleration during the time step and of analytical integration of acceleration and velocity to step forward from the initial to the final condition for each time step.

In general, these methods are based on the two equations

$$\dot{x}_1 = \dot{x}_0 + \int_0^h \ddot{x}(\tau) d\tau,$$
$$x_1 = x_0 + \int_0^h \dot{x}(\tau) d\tau,$$

which express the final velocity and the final displacement in terms of the initial values  $x_0$  and  $\dot{x}_0$  and some definite integrals that depend on the *assumed* variation of the acceleration during the time step.

# Integration Methods

Depending on the different assumption we can make on the variation of velocity, different integration methods can be derived.

We will see



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Depending on the different assumption we can make on the variation of velocity, different integration methods can be derived.

We will see

- the constant acceleration method,
- the linear acceleration method,
- the family of methods known as *Newmark Beta Methods*, that comprises the previous methods as particular cases.

## Constant Acceleration

Here we assume that the acceleration is constant during each time step, equal to the mean value of the initial and final values:

$$\ddot{x}(\tau) = \ddot{x}_0 + \Delta\ddot{x}/2,$$

where  $\Delta\ddot{x} = \ddot{x}_1 - \ddot{x}_0$ , hence

$$\dot{x}_1 = \dot{x}_0 + \int_0^h (\ddot{x}_0 + \Delta\ddot{x}/2) d\tau$$

$$\Rightarrow \Delta\dot{x} = \ddot{x}_0 h + \Delta\ddot{x} h/2$$

$$x_1 = x_0 + \int_0^h (\dot{x}_0 + (\ddot{x}_0 + \Delta\ddot{x}/2)\tau) d\tau$$

$$\Rightarrow \Delta x = \dot{x}_0 h + (\ddot{x}_0) h^2/2 + \Delta\ddot{x} h^2/4$$

## Constant acceleration

Taking into account the two equations on the right of the previous slide, and solving for  $\Delta\dot{x}$  and  $\Delta\ddot{x}$  in terms of  $\Delta x$ , we have

$$\Delta\dot{x} = \frac{2\Delta x - 2h\dot{x}_0}{h}, \quad \Delta\ddot{x} = \frac{4\Delta x - 4h\dot{x}_0 - 2\ddot{x}_0h^2}{h^2}.$$

We have two equations and three unknowns... Assuming that the system characteristics are constant during a single step, we can write the equation of motion at times  $\tau = h$  and  $\tau = 0$ , subtract member by member and write the *incremental equation of motion*

$$m\Delta\ddot{x} + c\Delta\dot{x} + k\Delta x = \Delta p,$$

that is a third equation that relates our unknowns.

## Constant acceleration

Substituting the above expressions for  $\Delta\dot{x}$  and  $\Delta\ddot{x}$  in the incremental eq. of motion and solving for  $\Delta x$  gives, finally,

$$\Delta x = \frac{\tilde{p}}{\tilde{k}}, \quad \Delta\dot{x} = \frac{2\Delta x - 2h\dot{x}_0}{h}$$

where

$$\tilde{k} = k + \frac{2c}{h} + \frac{4m}{h^2}$$
$$\tilde{p} = \Delta p + 2c\dot{x}_0 + m\left(2\ddot{x}_0 + \frac{4}{h}\dot{x}_0\right)$$

While it is possible to compute the final acceleration in terms of  $\Delta x$ , to achieve a better accuracy it is usually computed solving the equation of equilibrium written at the end of the time step.

## Constant Acceleration

Two further remarks

1. The method is *unconditionally stable*
2. The effective stiffness, disregarding damping, is  $\tilde{k} \approx k + 4m/h^2$ .

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2. The effective stiffness, disregarding damping, is  $\tilde{k} \approx k + 4m/h^2$ .

Dividing both members of the above equation by  $k$  it is

$$\frac{\tilde{k}}{k} = 1 + \frac{4}{\omega_n^2 h^2} = 1 + \frac{4}{(2\pi/T_n)^2 h^2} = \frac{T_n^2}{\pi^2 h^2},$$

The number  $n_T$  of time steps in a period  $T_n$  is related to the time step duration,  $n_T = T_n/h$ , solving for  $h$  and substituting in our last equation, we have

$$\frac{\tilde{k}}{k} \approx 1 + \frac{n_T^2}{\pi^2}$$

For, e.g.,  $n_T = 2\pi$  it is  $\tilde{k}/k \approx 1 + 4$ , the mass contribution to the effective stiffness is four times the elastic stiffness and the 80% of the total.



## Linear Acceleration

We assume that the acceleration is linear, i.e.

$$\ddot{x}(t) = \ddot{x}_0 + \Delta\ddot{x}\frac{\tau}{h}$$

hence

$$\Delta\dot{x} = \ddot{x}_0 h + \Delta\ddot{x}h/2, \quad \Delta x = \dot{x}_0 h + \ddot{x}_0 h^2/2 + \Delta\ddot{x}h^2/6$$

Following a derivation similar to what we have seen in the case of constant acceleration, we can write, again,

$$\Delta x = \left(k + 3\frac{c}{h} + 6\frac{m}{h^2}\right)^{-1} \left[\Delta p + c\left(\ddot{x}_0\frac{h}{2} + 3\dot{x}_0\right) + m\left(3\ddot{x}_0 + 6\frac{\dot{x}_0}{h}\right)\right]$$

$$\Delta\dot{x} = \Delta x\frac{3}{h} - 3\dot{x}_0 - \ddot{x}_0\frac{h}{2}$$

## Linear Acceleration

The linear acceleration method is *conditionally stable*, the stability condition being

$$\frac{h}{T} \leq \frac{\sqrt{3}}{\pi} \approx 0.55$$

When dealing with SDOF systems, this condition is never of concern, as we need a shorter step to accurately describe the response of the oscillator, let's say  $h \leq 0.12T$ ...

When stability is not a concern, the accuracy of the linear acceleration method is far superior to the accuracy of the constant acceleration method, so that this is the method of choice for the analysis of SDOF systems.

## Newmark Beta Methods

The constant and linear acceleration methods are just two members of the family of Newmark Beta methods, where we write

$$\begin{aligned}\Delta \dot{x} &= (1 - \gamma)h\ddot{x}_0 + \gamma h\ddot{x}_1 \\ \Delta x &= h\dot{x}_0 + \left(\frac{1}{2} - \beta\right)h^2\ddot{x}_0 + \beta h^2\ddot{x}_1\end{aligned}$$

The factor  $\gamma$  weights the influence of the initial and final accelerations on the velocity increment, while  $\beta$  has a similar role with respect to the displacement increment.

## Newmark Beta Methods

Using  $\gamma \neq 1/2$  leads to numerical damping, so when analysing SDOF systems, one uses  $\gamma = 1/2$  (numerical damping may be desirable when dealing with MDOF systems).

Using  $\beta = \frac{1}{4}$  leads to the constant acceleration method, while  $\beta = \frac{1}{6}$  leads to the linear acceleration method. In the context of MDOF analysis, it's worth knowing what is the minimum  $\beta$  that leads to an unconditionally stable behaviour.

## Newmark Beta Methods

The general format for the solution of the incremental equation of motion using the Newmark Beta Method can be written as follows:

$$\Delta x = \frac{\Delta \tilde{p}}{\tilde{k}}$$
$$\Delta v = \frac{\gamma}{\beta} \frac{\Delta x}{h} - \frac{\gamma}{\beta} v_0 + h \left( 1 - \frac{\gamma}{2\beta} \right) a_0$$

with

$$\tilde{k} = k + \frac{\gamma}{\beta} \frac{c}{h} + \frac{1}{\beta} \frac{m}{h^2}$$
$$\Delta \tilde{p} = \Delta p + \left( h \left( \frac{\gamma}{2\beta} - 1 \right) c + \frac{1}{2\beta} m \right) a_0 + \left( \frac{\gamma}{\beta} c + \frac{1}{\beta} \frac{m}{h} \right) v_0$$

## Non Linear Systems

A convenient procedure for integrating the response of a non linear system is based on the incremental formulation of the equation of motion, where for the stiffness and the damping were taken values representative of their variation during the time step: in line of principle, the mean values of stiffness and damping during the time step, or, as this is usually not possible, their initial values,  $k_0$  and  $c_0$ .

The Newton-Raphson method can be used to reduce the unbalanced forces at the end of the step.

## Modified Newton-Raphson Method

Usually we use the modified Newton-Raphson method, characterised by not updating the system stiffness at each iteration. In pseudo-code, referring for example to the Newmark Beta Method

```
x1,v1,f1 = x0,v0,f0 % initialisation; gb=gamma/beta
```

```
Dr = DpTilde
```

```
loop:
```

```
  Dx = Dr/kTilde
```

```
  x2 = x1 + Dx
```

```
  v2 = gb*Dx/h - gb*v1 + (1-gb/2)*h*a0
```

```
  x_pl = update_u_pl(...)
```

```
  f2 = k*(x2-x_pl)
```

```
  % important
```

```
  Df = (f2-f1) + (kTilde-k_ini)*Dx
```

```
  Dr = Dr - Df
```

```
x1 = x2; v1 = v2; f1 = f2
```

## Exercise

A system has a mass  $m = 1000\text{kg}$ , a stiffness  $k = 40000\text{N/m}$  and a viscous damping whose ratio to the critical damping is  $\zeta = 0.03$ .

The spring is elastoplastic, with a yielding force of  $2500\text{N}$ .

The load is an half-sine impulse, with duration  $0.3\text{s}$  and maximum value of  $6000\text{N}$ .

Use the constant acceleration method to integrate the response, with  $h = 0.05\text{s}$  and, successively,  $h = 0.02\text{s}$ . Note that the stiffness is either 0 or  $k$ , write down the expression for the effective stiffness and loading in the incremental formulation, write a spreadsheet or a program to make the computations.